Rhea Scatterometry Rev 27

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- Sequence: s22
- Rev: 027
- Observation Id: rh_027_1
- Target Body: Rhea

1 Introduction

This memo describes one of the Cassini RADAR activities for the s22 sequence of the Saturn Tour. A sequence design memo provides the science context of the scheduled observations, an overview of the pointing design, and guidlines for preparing the RADAR IEB.

This IEB is for the fourth Rhea distant scatterometer observation. A 3-hour warmup occurs first using the parameters shown in table 4. This observation splits a data allocation of about 300 Mbits with the nearby Dione-27 observation. This observations has exactly the same structure as the Dione-027 observation, and this memo is the same except for computed and plotted values.

2 CIMS and Division Summary

CIMS ID	Start	End	Duration	Comments
027OT_WARM4RH001_RIDER	2006-229T03:25:00	2006-229T05:15:00	01:50:0.0	Warmup for scatterometry and simultaneous radiometry of icy satellite.
027RH_SCATTRADL001_PRIME	2006-229T05:15:00	2006-229T06:50:00	01:35:0.0	Point -Z axis at target and execute raster scan(s) centered on target. Obtain simultaneous scatterometry and radiometry.

Table 1: rh_027_1 CIMS Request Sequence

Division	Name	Start	Duration	Data Vol	Comments
a	distant_radiometer	00:00:0.0	01:40:0.0	1.5	Warmup
b	distant_radiometer	01:40:0.0	00:26:0.0	1.5	Warmup
c	scat_compressed	02:06:0.0	00:25:30.0	6.1	Scatterometer on/off-target rcv only compressed attenuator tests
d	distant_scatterometer	02:31:30.0	00:04:30.0	43.2	Scatterometer target-center stare with tone
e	distant_scatterometer	02:36:0.0	00:10:0.0	96.0	Scatterometer target-center stare with tone
f	scat_compressed	02:46:0.0	00:24:0.0	5.8	Scatterometer on/off-target rcv only compressed
g	distant_radiometer	03:10:0.0	00:10:0.0	0.6	Closing radiometry
Total				154.7	

Table 2: Division summary. Data volumes (Mbits) are estimated from maximum data rate and division duration.

Div	Alt (km)	Slant range (km)	B3 Size (target dia)	B3 Dop. Spread (Hz)
a	239449	off target	1.02	off target
b	198686	off target	0.84	off target
c	190214	off target	0.81	off target
d	182872	182872	0.78	3030
e	181681	181681	0.77	3049
f	179150	179150	0.76	3089
g	173748	off target	0.74	off target

Table 3: Division geometry summary. Values are computed at the start of each division. B3 Doppler spread is for two-way 3-dB pattern. B3 size is the one-way 3-dB beamwidth

Name	Nominal	a	b	Mismatch	Comments
mode	radiometer	radiometer	radiometer	no	
start_time (min)	varies	0.0	100.0	no	
end_time (min)	varies	100.0	126.0	no	
time_step (s)	varies	1800.0	1800.0	no	Used by radiometer only modes
bem	00100	00100	00100	no	
baq	don't care	5	5	no	
csr	6	6	6	no	
noise_bit_setting	don't care	4.0	4.0	no	
dutycycle	don't care	0.38	0.38	no	
prf (Hz)	don't care	1000	1000	no	
tro	don't care	6	6	no	
number_of_pulses	don't care	8	8	no	
n_bursts_in_flight	don't care	1	1	no	
percent_of_BW	don't care	100.0	100.0	no	
auto_rad	on	on	on	no	
rip (ms)	34.0	34.0	34.0	no	starting value for auto-rad
max_data_rate	1.000	0.248	0.992	yes	1 Kbps - 1 s burst period which is adequate for slow radiometer scans
interleave_flag	off	off	off	no	
interleave_duration (min)	don't care	10.0	10.0	no	

Table 4: rh_027_1 div_ab distant_radiometer block

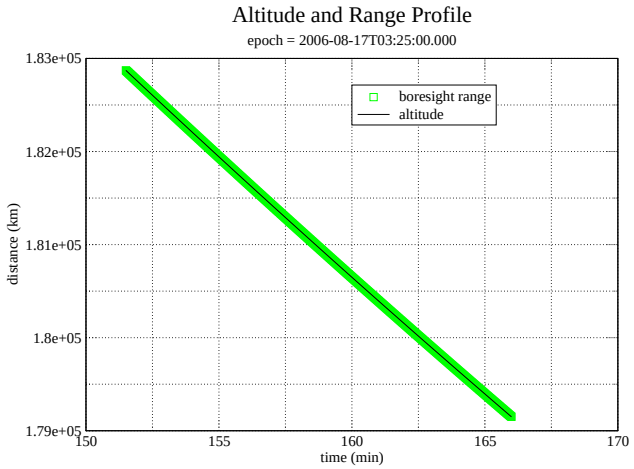

Figure 1: Div's D,E: Altitude and range to the boresight point

Each RADAR observation is represented to the project by a set of requests in the Cassini Information Management System (CIMS). The CIMS database contains requests for pointing control, time, and data volume. The CIMS requests show a high-level view of the sequence design.

The CIMS requests form the basis of a pointing design built using the project pointing design tool (PDT). The details of the pointing design are shown by the PDT plots on the corresponding tour sequence web page. (See https://cassini.jpl.nasa.gov/radar.) The RADAR pointing sequence is ultimately combined with pointing sequences from other instruments to make a large merged c-kernel. C-kernels are files containing spacecraft attitude data.

A RADAR tool called RADAR Mapping and Sequencing Software (RMSS) reads the merged c-kernel along with other navigation data files, and uses these data to produce a set of instructions for the RADAR observation. The RADAR instructions are called an Instrument Execution Block (IEB). The IEB is produced by running RMSS with a radar config file that controls the process of generating IEB instructions for different segments of time. These segments of time are called divisions with a particular behavior defined by a set of division keywords in the config file. Table 2 shows a summary of the divisions used in this observation. Subsequent sections will show and discuss the keyword selections made for each division. Each division table shows a set of nominal parameters that are determined by the operating mode (eg., distant scatterometry, SAR low-res inbound). The actual division parameters from the config file are also shown, and any meaningful mismatches are flagged.

3 Div's D,E: Rhea Scatterometry

Figures 1 and 2 show the pointing design for the scatterometry stare from the merged ckernel. The angular size of the target is about 8.3 mrad during this division. The beam 3 beamwidth is 6 mrad.

The division parameters for the radiometer segments are shown in table 4. The division parameters for the compressed scatterometer receive only integrations are shown in table 5 and the tone target integration divisions are shown in table 6.

3.1 Scatterometer Receive Only Measurements

Div C is a place holder for manually placed receive only data that will survey attenuator settings in various modes. These blocks of instructions are placed in distant icy satellite observations if data volume and pointing permit. They will improve calibration of the radar in all modes. Div C covers the turn onto the target so there is off-target and on-target receive only data available. Div F covers the end of the stare and provides more receive only integration using scatterometer mode and the 9 dB attenuator setting used in the tone integrations. All of the receive only data is collected in compressed mode to get more integration time. The division PRF and number of pulses (1202 Hz and 160 respectively) are chosen to fill the science data buffer. These parameters give the best performance possible from the

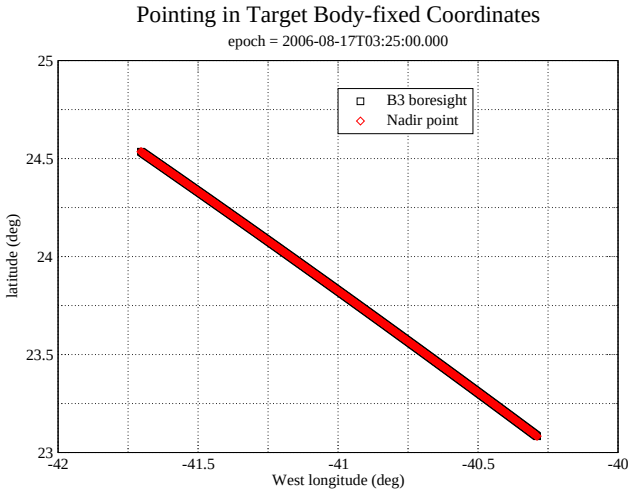

Figure 2: Div's D,E: Stare in target body-fixed coordinates

Name	Nominal	c	f	Mismatch	Comments
mode	scat_compressed	scat_compressed	scat_compressed	yes	
start_time (min)	varies	126.0	166.0	no	
end_time (min)	varies	151.5	190.0	no	
time_step (s)	don't care	20.0	20.0	no	Set by valid time calculation
bem	00100	00100	00100	no	
baq	3	3	3	no	3 - PRI summation
csr	1	1	1	no	1 - receive only antenna measurement
noise_bit_setting	4.0	4.0	4.0	no	9 dB setting used by all low SNR scatterometry
dutycycle	0.70	0.70	0.70	no	
prf (Hz)	1200	1200	1200	no	
tro	don't care	6	6	no	automatically set to 6
number_of_pulses	150	150	150	no	Set with the PRF to fill the science data buffer - Only 2 PRI's worth of data are downlinked.
n_bursts_in_flight	1	1	1	no	
percent_of_BW	100.0	0.0	0.0	yes	
auto_rad	on	on	on	no	
rip (ms)	34.0	34.0	34.0	no	
max_data_rate	8.000	4.000	4.000	yes	
interleave_flag	off	off	off	no	
interleave_duration (min)	don't care	10.0	10.0	no	

Table 5: rh_027_1 div_cf scat_compressed block

Name	Nominal	d	e	Mismatch	Comments
mode	scatterometer	scatterometer	scatterometer	no	
start_time (min)	varies	151.5	156.0	no	
end_time (min)	varies	156.0	166.0	no	
time_step (s)	don't care	8.0	8.0	no	Used when BIF > 1, otherwise set by valid time calculation
bem	00100	00100	00100	no	
baq	5	5	5	no	
csr	0	0	0	no	0 - normal operation with fixed attenuator set to match Phoebe for easier cross-calibration
noise_bit_setting	4.0	4.0	4.0	no	Scat signal set higher than ALT/SAR
dutycycle	0.70	0.70	0.70	no	
prf (Hz)	varies	5435	5435	no	Set to cover doppler spread and to allow CSF = integer multiple
tro	6	6	6	no	6 - allows for some noise only data in time domain
number_of_pulses	varies	230	230	no	depends on PRF choice (can have more shorter pulses)
n_bursts_in_flight	varies	3	3	no	Used to increase PRF and data rate at long range
percent_of_BW	0.0	0.0	0.0	no	
auto_rad	on	on	on	no	
rip (ms)	34.0	34.0	34.0	no	
max_data_rate	200.000	160.000	160.000	yes	Kbps - determines burst period
interleave_flag	off	off	off	no	
interleave_duration (min)	don't care	10.0	10.0	no	

Table 6: rh_027_1 div_de distant_scatterometer block

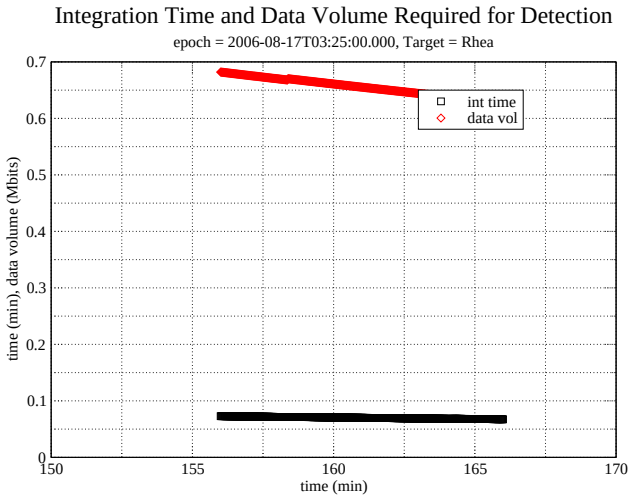

Figure 3: Scatterometry Div E: Detection integration time required for a single point detection using optimal chirp bandwidth

compressed mode.

3.2 Scatterometer Performance

The detection performance is shown in figures 3, 4, and 5. Figure 5 shows that range processing not possible due to high K_{pc}. Disk integrated results should be very stable.

The maximum doppler spread in Div e is 3089 Hz which comes from rotation and spacecraft motion. The PRF needs to be higher than the doppler spread to support potential range-doppler processing, and is set by division parameter to 5435 Hz. With this PRF, the range amiguity spacing is 28 km while Rhea is 764 km in radius. The range-spread of the beam depends on where it is pointed. For target centered pointing the cosine law can be applied to solve the geometry. At 181681 km range, the range-spread is 240 km.

4 Div G: Rhea Radiometry

This observation and the adjacent Rhea observation do not have radiometry scans due to insufficient time. Radiometry data is always collected even during scatterometer mode, so there will be on-target and off-target radiometry data that can be used to determine the disk brightness temperature.

5 Revision History

1. Jun 15, 2006: Initial Release

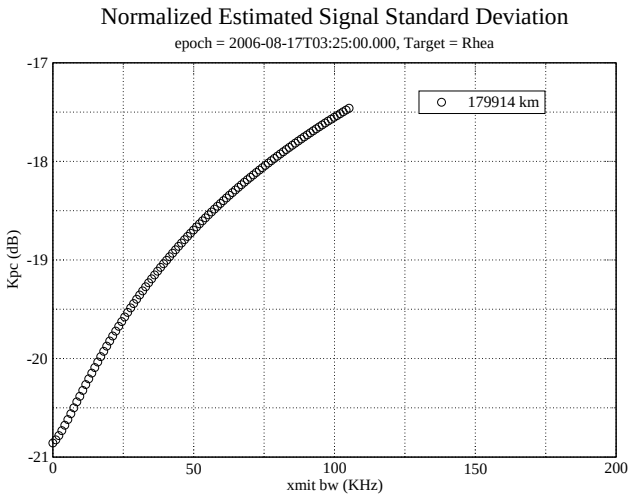

Figure 4: Outbound observation Div E: Normalized estimated signal standard deviation for a disk integrated observation using optimal chirp bandwidth and assuming all the bursts occur at minimum range, and 15 minutes away from minimum range.

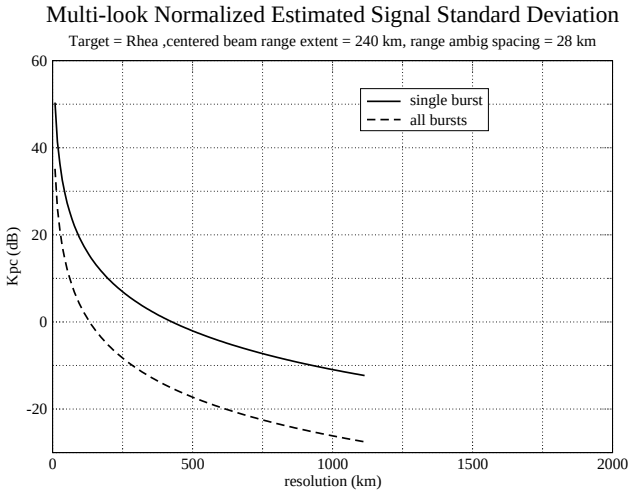

Figure 5: Outbound observation Div E: Normalized estimated signal standard deviation for a range/doppler cell as a function of resolution. Range/doppler resolution elements are both set equal to the specified resolution. Results are shown for a single burst, and for all the bursts in this division. Calculations are performed using the geometry at the start of the division. The presence of ambiguities are not shown.

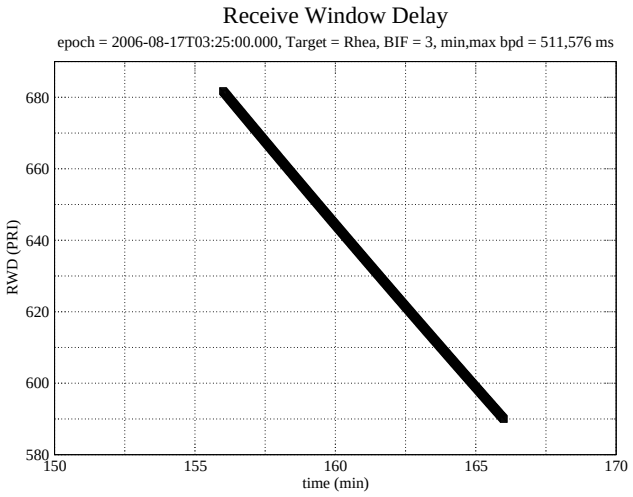

Figure 6: Div E: Inbound scatterometer receive window delay. Subtitle shows the minimum and maximum burst periods that are in principle compatible with the division selected number of bursts in flight.

6 Acronym List

AL	Acronym List
ALT	Altimeter - one of the radar operating modes
BAQ	Block Adaptive Quantizer
CIMS	Cassini Information Management System - a database of observations
Ckernel	NAIF kernel file containing attitude data
DLAP	Desired Look Angle Profile - spacecraft pointing profile designed for optimal SAR performance
ESS	Energy Storage System - capacitor bank used by RADAR to store transmit energy
IEB	Instrument Execution Block - instructions for the instrument
ISS	Imaging Science Subsystem
IVD	Inertial Vector Description - attitude vector data
IVP	Inertial Vector Propagator - spacecraft software, part of attitude control system
INMS	Inertial Neutral Mass Spectrometer - one of the instruments
NAIF	Navigation and Ancillary Information Facility
ORS	Optical Remote Sensing instruments
PDT	Pointing Design Tool
PRI	Pulse Repetition Interval
PRF	Pulse Repetition Frequency
RMSS	Radar Mapping Sequencing Software - produces radar IEB's
SAR	Synthetic Aperture Radar - radar imaging mode
SNR	Signal to Noise Ratio
SOP	Science Operations Plan - detailed sequence design
SOPUD	Science Operations Plan Update - phase of sequencing when SOP is updated prior to actual sequencing
SSG	SubSequence Generation - spacecraft/instrument commands are produced
SPICE	Spacecraft, Instrument, C-kernel handling software - supplied by NAIF to use NAIF kernel files.
TRO	Transmit Receive Offset - round trip delay time in units of PRI