

RADAR Titan Flyby during S73/T83

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- Sequence: *s73*
- Rev: 166
- Observation Id: t83
- Target Body: Titan
- Data Take Number: 239
- PDT Config File: *S73_sip_port3_120202_pdt.cfg*
- SMT File: *S73_120130.smt*
- PEF File: *z0730b.pef*

1 Introduction

This memo describes the Cassini RADAR activities for the T83 Titan flyby. This SAR data collection occurs during the S73 sequence of the Saturn Tour. This is a partial radar pass with ride-along SAR inbound imaging and normal outbound imaging. A sequence design memo provides the science context of the scheduled observations, an overview of the pointing design, and guidelines for preparing the RADAR IEB.

2 CIMS and Division Summary

Each RADAR observation is represented to the project by a set of requests in the Cassini Information Management System (CIMS). The CIMS database contains requests for pointing control, time, and data volume. The CIMS requests show a high-level view of the sequence design. Table 1 shows the CIMS request summary for this observation. Although the CIMS requests show Low-SAR intervals, in reality the radar will be operated in Hi-SAR mode through most of this flyby.

The CIMS requests form the basis of a pointing design built using the project pointing design tool (PDT). The details of the pointing design are shown by the PDT plots on the corresponding tour sequence web page. (See <https://cassini.jpl.nasa.gov/radar>.) The RADAR pointing sequence is ultimately combined with pointing sequences from other instruments to make a large merged c-kernel. C-kernels are files containing spacecraft attitude data.

A RADAR tool called RADAR Mapping and Sequencing Software (RMSS) reads the merged c-kernel along with other navigation data files, and uses these data to produce a set of instructions for the RADAR observation. The RADAR instructions are called an Instrument Execution Block (IEB). The IEB is produced by running RMSS with a radar config file that controls the process of generating IEB instructions for different segments of time. These segments of time are called divisions with a particular behavior defined by a set of division keywords in the config file. Table 2 shows a summary of the divisions used in this observation. Table 3 shows a summary of some key geometry values for each division.

CIMS ID	Start	End	Duration	Comments
166TLT83WARMUP001_RIDER	2012-142T19:43:11	2012-142T22:43:11	03:00:0.0	
166TLT83INSCAT001_PRIME	2012-142T22:43:11	2012-142T23:58:11	01:15:0.0	
166TLT83IHISAR001_PRIME	2012-142T23:58:11	2012-143T00:39:11	00:41:0.0	
166TLT83INALT001_PRIME	2012-143T00:40:11	2012-143T00:55:11	00:15:0.0	
166TLT83IRASAR001_PRIME	2012-143T00:55:11	2012-143T01:05:11	00:10:0.0	
166TLT83IRASAR002_PRIME	2012-143T01:05:11	2012-143T01:10:11	00:05:0.0	
166TLT83OUTSAR001_PRIME	2012-143T01:10:11	2012-143T01:28:11	00:18:0.0	
166TLT83OUTALT001_PRIME	2012-143T01:28:11	2012-143T01:40:11	00:12:0.0	
166TLT83OHISAR001_PRIME	2012-143T01:40:11	2012-143T02:00:11	00:20:0.0	
166TLT83OUTSCT001_PRIME	2012-143T02:22:11	2012-143T03:25:11	01:03:0.0	
166TLT83OUTRAD001_PRIME	2012-143T03:25:11	2012-143T07:10:11	03:45:0.0	

Table 1: t83 CIMS Request Sequence

3 Overview

T83 is a partial pass. The observation starts with scatterometry scans, and then switching to high altitude imaging. Inbound high-altitude imaging consists of 8 scan lines and covers a region overlapping part of the T3 SAR swath. A standard inbound altimeter segment follows the inbound high-altitude imaging. Near closest approach, the spacecraft turns to put -X in the direction of motion for INMS, while RADAR collects ride-along imaging. Starting at close approach, the spacecraft switches to a normal SAR imaging profile and the RADAR collects a standard outbound SAR swath. A standard outbound altimeter segment follows, and then an outbound high altitude imaging segment with 7 scan lines that overlap the T43 swath. After the high-altitude imaging, an extra altimeter segment is collected while the spacecraft transitions from thrusters to wheel attitude control. Then an outbound scatterometer scan is performed followed by two radiometry raster scans with orthogonal polarizations. At either end of close approach altimetry, two short atmospheric probe observations are inserted.

4 Mode Specific Operation and Performance

Many details of standard radar sequencing during the 4 main modes (Radiometry, Scatterometry, Altimetry, and SAR) have been discussed in previous sequence memos for prior observations. Refer to these for details. Some select performance highlights are illustrated in figures and explained in the following subsections.

4.1 SAR Resolution Performance

For all of the SAR divisions the effective resolution can be calculated from the same equations used in the high-altitude imaging discussion. Figure 1 shows the results from these equations using the parameters from the IEB as generated by RMSS. The calculations are performed for the boresight of beam 3 which is the center of the swath.

Projected range increases with decreasing incidence angle, so the range resolution varies across the swath with better resolution at the outer edge. The SAR pointing profile decreases the incidence angle as time progresses and altitude increases, so there is progressive deterioration of range resolution away from closest approach. The projected range resolution rapidly deteriorates as the incidence angle decreases toward zero at the very beginning and end of the swath.

Azimuth resolution is a function of the synthetic aperture size which is determined by the length of the receive window in each burst (assuming the receive window is always filled with echos). Azimuth resolution deteriorates less quickly because the number of pulses and the length of the receive window are increased as altitude increases which mitigates the increasing doppler bandwidth of the beam patterns. The receive window length increases to fill the round trip time until the science data buffer is filled. At this point it is no longer possible to extend the receive window, and azimuth resolution starts to deteriorate more rapidly.

Division	Name	Start	Duration	Data Vol	Comments
a	Warmup	-5:30:0.0	03:23:0.0	12.1	Warmup
b	standard_radiometer_inbound	-2:07:0.0	00:01:0.0	0.1	radiometry fill
c	standard_scatterometer_inbound	-2:06:0.0	00:51:0.0	91.8	Inbound scatterometry during turn to scatterometer imaging
d	scatterometer_imaging	-1:15:0.0	00:25:0.0	82.5	Inbound scatterometer imaging
e	scatterometer_imaging	-0:50:0.0	00:21:0.0	69.3	Inbound scatterometer imaging
f	standard_altimeter_inbound	-0:29:0.0	00:09:12.0	19.9	Inbound altimetry
g	standard_scatterometer_inbound	-0:19:48.0	00:00:4.0	0.6	Atmospheric Probe with Chirp
h	standard_scatterometer_inbound	-0:19:44.0	00:00:2.0	0.3	Atmospheric Probe with Tone
i	standard_sar_hi	-0:19:42.0	00:01:30.0	5.4	SAR Turn transition, beam 3 only
j	standard_radiometer_inbound	-0:18:12.0	00:12:12.0	0.7	Inbound radiometry
k	standard_sar_hi	-0:06:0.0	00:01:0.0	3.6	SAR Turn transition, beam 3 only
l	standard_sar_hi	-0:05:0.0	00:21:0.0	294.8	Inbound standard Hi-SAR
m	standard_sar_pingpong	00:16:0.0	00:02:36.0	36.5	Inbound SAR ping-pong
n	standard_sar_hi	00:18:36.0	00:00:42.0	2.5	SAR Turn transition, beam 3 only
o	standard_scatterometer_outbound	00:19:18.0	00:00:4.0	0.6	Atmospheric Probe with Tone
p	standard_scatterometer_outbound	00:19:22.0	00:00:2.0	0.3	Atmospheric Probe with Chirp
q	standard_altimeter_outbound	00:19:24.0	00:08:48.0	19.0	Outbound altimetry
r	scatterometer_imaging	00:28:12.0	00:09:48.0	32.3	Outbound scatterometer imaging
s	scatterometer_imaging	00:38:0.0	00:12:0.0	39.6	Outbound scatterometer imaging
t	standard_altimeter_outbound	00:50:0.0	00:22:0.0	42.2	Outbound altimetry
u	standard_scatterometer_outbound	01:12:0.0	01:02:0.0	111.6	Outbound scatterometry
v	standard_radiometer_outbound	02:14:0.0	03:46:0.0	13.5	Outbound radiometry
Total				879.1	

Table 2: Division summary. Data volumes (Mbits) are estimated from maximum data rate and division duration.

Div	Alt (km)	Slant range (km)	B3 Size (target dia)	B3 Dop. Spread (Hz)
a	106209	off target	0.14	off target
b	39644	off target	0.05	off target
c	39317	off target	0.05	off target
d	22645	23079	0.03	373
e	14514	15590	0.02	535
f	7798	7807	0.01	870
g	4983	4983	0.01	1190
h	4963	4963	0.01	1193
i	4953	4954	0.01	1194
j	4513	5590	0.01	1268
k	1500	2136	0.01	2200
l	1342	1751	0.00	2289
m	3884	3963	0.01	1391
n	4630	4686	0.01	1247
o	4835	4835	0.01	1213
p	4855	4855	0.01	1210
q	4864	4864	0.01	1208
r	7548	7548	0.01	891
s	10651	11164	0.02	685
t	14515	14515	0.02	535
u	21668	21668	0.03	386
v	41936	43480	0.06	241

Table 3: Division geometry summary. Values are computed at the start of each division. B3 Doppler spread is for two-way 3-dB pattern. B3 size is the one-way 3-dB beamwidth

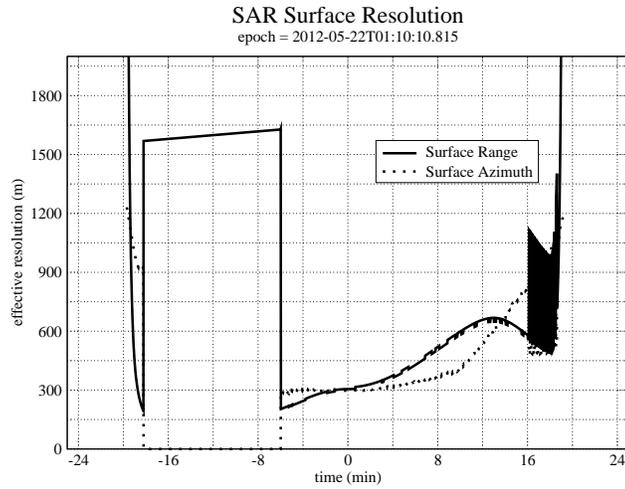


Figure 1: SAR projected range and azimuth resolution. These values are computed from the IEB parameters and are not related to the pixel size in the BIDR file. The pixel size was selected to be always smaller than the real resolution.

5 Revision History

1. March 13, 2013: Final release

6 Acronym List

ALT	Altimeter - one of the radar operating modes
BAQ	Block Adaptive Quantizer
CIMS	Cassini Information Management System - a database of observations
Ckernel	NAIF kernel file containing attitude data
DLAP	Desired Look Angle Profile - spacecraft pointing profile designed for optimal SAR performance
ESS	Energy Storage System - capacitor bank used by RADAR to store transmit energy
IEB	Instrument Execution Block - instructions for the instrument
ISS	Imaging Science Subsystem
IVD	Inertial Vector Description - attitude vector data
IVP	Inertial Vector Propagator - spacecraft software, part of attitude control system
INMS	Inertial Neutral Mass Spectrometer - one of the instruments
NAIF	Navigation and Ancillary Information Facility
ORS	Optical Remote Sensing instruments
PDT	Pointing Design Tool
PRI	Pulse Repetition Interval
PRF	Pulse Repetition Frequency
RMSS	Radar Mapping Sequencing Software - produces radar IEB's
SAR	Synthetic Aperture Radar - radar imaging mode
SNR	Signal to Noise Ratio
SOP	Science Operations Plan - detailed sequence design
SOPUD	Science Operations Plan Update - phase of sequencing when SOP is updated prior to actual sequencing
SSG	SubSequence Generation - spacecraft/instrument commands are produced
SPICE	Spacecraft, Instrument, C-kernel handling software - supplied by NAIF to use NAIF kernel files.
TRO	Transmit Receive Offset - round trip delay time in units of PRI