Dione Scatterometry Rev 050

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- Sequence: s34
- Rev: 050
- Observation Id: di_050_1
- Target Body: Dione

1 Introduction

This memo describes one of the Cassini RADAR activities for the s34 sequence of the Saturn Tour. A sequence design memo provides the science context of the scheduled observations, an overview of the pointing design, and guidelines for preparing the RADAR IEB. A 3-hour warmup occurs first using the parameters shown in table 4.

2 CIMS and Division Summary

CIMS ID	Start	End	Duration	Comments
050OT_WARMUP4DI001_RIDER	2007-272T21:00:00	2007-273T00:00:00	03:00:0.0	Warmup for scat-
				terometry and simul-
				taneous radiometry
				of icy satellite.
050DI_SCATTRAD001_PRIME	2007-273T00:00:00	2007-273T03:00:00	03:00:0.0	Point -Z axis at
				target and execute
				raster scan(s) cen-
				tered on target.
				Obtain simultaneous
				scatterometry and
				radiometry.

Table 1: di_050_1 CIMS Request Sequence

Each RADAR observation is represented to the project by a set of requests in the Cassini Information Management System (CIMS). The CIMS database contains requests for pointing control, time, and data volume. The CIMS requests show a high-level view of the sequence design.

The CIMS requests form the basis of a pointing design built using the project pointing design tool (PDT). The details of the pointing design are shown by the PDT plots on the corresponding tour sequence web page. (See

Division	Name	Start	Duration	Data Vol	Comments	
а	distant_radiometer	00:00:0.0	03:10:0.0	11.3	Warmup	
b	distant_radiometer	03:10:0.0	00:05:0.0	0.3	Warmup	
с	distant_radiometer	03:15:0.0	00:59:0.0	3.5	Radiometry Raster Scan	
d	distant_radiometer	04:14:0.0	00:08:0.0	0.5	Radiometry	
e	distant_scatterometer	04:22:0.0	00:03:0.0	35.1	Corner 1 Scatterometer	
					stare with tone	
f	distant_radiometer	04:25:0.0	00:12:0.0	0.7	Radiometer during turn	
					transition	
g	distant_scatterometer	04:37:0.0	00:03:0.0	36.0	Corner 2 Scatterometer	
					stare with tone	
h	distant_radiometer	04:40:0.0	00:12:0.0	0.7	Radiometer during turn	
					transition	
i	distant_scatterometer	04:52:0.0	00:03:0.0	36.0	Corner 3 Scatterometer	
					stare with tone	
j	distant_radiometer	04:55:0.0	00:12:0.0	0.7	Radiometer during turn	
					transition	
k	distant_scatterometer	05:07:0.0	00:03:0.0	36.0	Corner 4 Scatterometer	
		0.5.40.0.0	00.45.0.0	0.0	stare with tone	
I	distant_radiometer	05:10:0.0	00:15:0.0	0.9	Radiometer during turn	
	1	05.05.0.0	00.04.0.0	40.0	transition	
m	distant_scatterometer	05:25:0.0	00:04:0.0	48.0	Scatterometer target-center	
	1:	05.20.0.0	00.06.0.0	72.0	stare with chirp	
n	distant_scatterometer	05:29:0.0	00:06:0.0	72.0	Scatterometer target-center	
	1	05.25.0.0	00.10.0.0	2.6	stare with tone	
0	distant_radiometer	05:35:0.0	00:10:0.0	2.6	kadiometer during turn	
		05.45.0.0	00.05.0.0	0.2		
р	scat_compressed	05:45:0.0	00:05:0.0	0.3	Scatterometer on/on-target	
Total				2016	rev only compressed	
Total				284.0		

Table 2: Division summary. Data volumes (Mbits) are estimated from maximum data rate and division duration.

Div	Alt (km)	Slant range (km)	B3 Size (target dia)	B3 Dop. Spread (Hz)
а	200864	off target	1.16	off target
b	123070	off target	0.71	off target
с	121295	off target	0.70	off target
d	101355	101432	0.59	2396
e	98794	98867	0.57	2386
f	97843	97914	0.57	2383
g	94085	94160	0.55	2370
h	93158	93231	0.54	2368
i	89497	89580	0.52	2359
j	88594	88675	0.51	2358
k	85034	85122	0.49	2353
1	84156	84243	0.49	2352
m	79847	79847	0.46	2353
n	78720	78720	0.46	2354
0	77048	77048	0.45	2357
р	74312	off target	0.43	off target

Table 3: Division geometry summary. Values are computed at the start of each division. B3 Doppler spread is for two-way 3-dB pattern. B3 size is the one-way 3-dB beamwidth

Name	Nominal	Actual	Mismatch	Comments
mode	radiometer	radiometer	no	
start_time (min)	varies	0.0	no	
end_time (min)	varies	190.0	no	
time_step (s)	varies	2700.0	no	Used by radiome-
				ter only modes
bem	00100	11111	yes	
baq	don't care	5	no	
csr	6	6	no	
noise_bit_setting	don't care	4.0	no	
dutycycle	don't care	0.38	no	
prf (Hz)	don't care	1000	no	
tro	don't care	0	no	
number_of_pulses	don't care	8	no	
n_bursts_in_flight	don't care	1	no	
percent_of_BW	don't care	100.0	no	
auto_rad	on	on	no	
rip (ms)	34.0	34.0	no	starting value for
				auto-rad
max_data_rate	0.992	0.992	no	1 Kbps - 1 s burst
				period which is
				adequate for slow
				radiometer scans
interleave_flag	off	off	no	
interleave_duration (min)	don't care	10.0	no	

Table 4: di_050_1 Div a distant_radiometer block

Name	Nominal	Actual	Mismatch	Comments
mode	radiometer	scat_compressed	yes	
start_time (min)	varies	335.0	no	
end_time (min)	varies	345.0	no	
time_step (s)	varies	12.0	no	Used by radiome-
				ter only modes
bem	00100	00100	no	
baq	don't care	3	no	
csr	6	1	yes	
noise_bit_setting	don't care	4.0	no	
dutycycle	don't care	0.38	no	
prf (Hz)	don't care	1200	no	
tro	don't care	6	no	
number_of_pulses	don't care	150	no	
n_bursts_in_flight	don't care	1	no	
percent_of_BW	don't care	0.0	no	
auto_rad	on	on	no	
rip (ms)	34.0	34.0	no	starting value for
				auto-rad
max_data_rate	0.992	4.300	yes	1 Kbps - 1 s burst
				period which is
				adequate for slow
				radiometer scans
interleave_flag	off	off	no	
interleave_duration (min)	don't care	10.0	no	

Table 5: di_050_1 Div o distant_radiometer
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https://cassini.jpl.nasa.gov/radar.) The RADAR pointing sequence is ultimately combined with pointing sequences from other instruments to make a large merged c-kernel. C-kernels are files containing spacecraft attitude data.

A RADAR tool called RADAR Mapping and Sequencing Software (RMSS) reads the merged c-kernel along with other navigation data files, and uses these data to produce a set of instructions for the RADAR observation. The RADAR instructions are called an Instrument Execution Block (IEB). The IEB is produced by running RMSS with a radar config file that controls the process of generating IEB instructions for different segments of time. These segments of time are called divisions with a particular behavior defined by a set of division keywords in the config file. Table 2 shows a summary of the division. Each division table shows a set of nominal parameters that are determined by the operating mode (eg., distant scatterometry, SAR low-res inbound). The actual division parameters from the config file are also shown, and any meaningful mismatches are flagged.

3 Receive Only Engineering Test Measurements

Div O (see table 5) provides scatt mode, 9 dB attenuator receive only data off and on target for calibration of the scatterometer tone integration data. All of the receive only data is collected in compressed mode to get more integration time. The PRF and number of pulses are chosen to fill the science data buffer. These parameters give the best performance possible from the compressed mode.

4 Div's E-N: Dione Scatterometry

Figures 1 and 2 show the pointing design for the scatterometry stare from the merged ckernel. The angular size of the target is about 11.1 mrad during this division. The beam 3 beamwidth is 6 mrad. The division parameters for the tone



Figure 1: Scatt: Altitude and range to the boresight point



Figure 2: Scatt: Stares in target body-fixed coordinates



Figure 3: Scatterometry Div M: Detection integration time required for a single point detection using optimal chirp bandwidth

target integration divisions are shown in table 6.

4.1 Scatterometer Performance

The detection performance is shown in figures 3, 4, and 5. The maximum doppler spread in Div n is 2357 Hz which comes from rotation and spacecraft motion. The PRF needs to be higher than the doppler spread to support potential range-doppler processing, and is set by division parameter to 2500 Hz. With this PRF, the range amiguity spacing is 60 km while Dione is 560 km in radius. The range-spread of the beam depends on where it is pointed. For target centered pointing the cosine law can be applied to solve the geometry. At 78720 km range, the range-spread is 55 km.

Figure 5 shows that range processing is marginal due to high K_{pc} . Since range ambiguities are not a problem, a chirp division is included to cover the possibility of range processing. Disk integrated results should be very stable.

5 Revision History

1. Jul 1, 2008: Initial Release

Name	Nominal	Actual	Mismatch	Comments
mode	scatterometer	scatterometer	no	
start_time (min)	varies	262.0	no	
end_time (min)	varies	265.0	no	
time_step (s)	don't care	6.0	no	Used when BIF >
				1, otherwise set
				by valid time cal-
				culation
bem	00100	00100	no	
baq	5	5	no	
csr	0	0	no	0 - normal op-
				eration with
				fixed attenuator
				set to match
				Phoebe for easier
				cross-calibration
noise_bit_setting	4.0	4.0	no	Scat signal set
				higher than
				ALT/SAR
dutycycle	0.70	0.70	no	
prf (Hz)	varies	4417	no	Set to cover
				doppler spread
				and to allow CSF
				= integer multiple
tro	6	6	no	6 - allows for
				some noise only
				data in time do-
				main
number_of_pulses	varies	190	no	depends on PRF
				choice (can have
				more shorter
				pulses)
n_bursts_in_flight	varies	2	no	Used to increase
				PRF and data rate
				at long range
percent_of_BW	0.0	0.0	no	
auto_rad	on	on	no	
rip (ms)	34.0	34.0	no	
max_data_rate	200.000	195.000	yes	Kbps - determines
				burst period
interleave_flag	off	off	no	
interleave_duration (min)	don't care	10.0	no	

Table 6: di_050_1 Div e distant_scatterometer block



Figure 4: Div M: Normalized estimated signal standard deviation for a disk integrated observation using optimal chirp bandwidth and assuming all the bursts occur at minimum range, and 15 minutes away from minimum range.



Figure 5: Div N: Normalized estimated signal standard deviation for a range/doppler cell as a function of resolution. Range/doppler resolution elements are both set equal to the specified resolution. Results are shown for a single burst, and for all the bursts in this division. Calculations are performed using the geometry at the start of the division. The presence of ambiguities are not shown.

6 Acronym List

ALT	Altimeter - one of the radar operating modes
BAQ	Block Adaptive Quantizer
CIMS	Cassini Information Management System - a database of observations
Ckernel	NAIF kernel file containing attitude data
DLAP	Desired Look Angle Profile - spacecraft pointing profile designed for optimal SAR performance
ESS	Energy Storage System - capacitor bank used by RADAR to store transmit energy
IEB	Instrument Execution Block - instructions for the instrument
ISS	Imaging Science Subsystem
IVD	Inertial Vector Description - attitude vector data
IVP	Inertial Vector Propagator - spacecraft software, part of attitude control system
INMS	Inertial Neutral Mass Spectrometer - one of the instruments
NAIF	Navigation and Ancillary Information Facility
ORS	Optical Remote Sensing instruments
PDT	Pointing Design Tool
PRI	Pulse Repetition Interval
PRF	Pulse Repetition Frequency
RMSS	Radar Mapping Sequencing Software - produces radar IEB's
SAR	Synthetic Aperture Radar - radar imaging mode
SNR	Signal to Noise Ratio
SOP	Science Operations Plan - detailed sequence design
SOPUD	Science Operations Plan Update - phase of sequencing when SOP is updated prior to actual sequencing
SSG	SubSequence Generation - spacecraft/instrument commands are produced
SPICE	Spacecraft, Instrument, C-kernel handling software - supplied by NAIF to use NAIF kernel files.
TRO	Transmit Receive Offset - round trip delay time in units of PRI